



SCHOOL OF MATHEMATICS AND STATISTICS

Spring Semester 2010–2011

Fluid Mechanics I

2 hours

*Answer **four** questions. You are advised **not** to answer more than four questions: if you do, only your best four will be counted.*

1 (i) Consider the following two flows:

(a)

$$\mathbf{u} = (-\Omega y, \Omega x, 0),$$

(b)

$$\mathbf{u} = \left( -\frac{\Omega y}{x^2 + y^2}, \frac{\Omega x}{x^2 + y^2}, 0 \right) \quad \text{for } x^2 + y^2 \neq 0,$$

where  $\Omega$  is constant. Compute the vorticity in each case and state which flow is rotational and which is irrotational. **(6 marks)**

(ii) For the above flows, draw stream lines for both (a) and (b). **(4 marks)**

(iii) The material derivative of vector  $\mathbf{A}$  is defined as

$$\frac{D\mathbf{A}}{Dt} = \frac{\partial \mathbf{A}}{\partial t} + (\mathbf{u} \cdot \nabla)\mathbf{A},$$

where  $\mathbf{u}$  denotes fluid velocity which is in general compressible.

Deduce the following formula:

$$\begin{aligned} \frac{D\mathbf{A}}{Dt} = \frac{\partial \mathbf{A}}{\partial t} + \frac{1}{2} [\nabla(\mathbf{u} \cdot \mathbf{A}) + (\nabla \times \mathbf{u}) \times \mathbf{A} \\ + (\nabla \times \mathbf{A}) \times \mathbf{u} - \nabla \times (\mathbf{u} \times \mathbf{A}) + \mathbf{u} \nabla \cdot \mathbf{A} - \mathbf{A} \nabla \cdot \mathbf{u}] \end{aligned}$$

You may use the following vector identities:

$$\nabla(\mathbf{A} \cdot \mathbf{B}) = (\mathbf{A} \cdot \nabla)\mathbf{B} + (\mathbf{B} \cdot \nabla)\mathbf{A} + \mathbf{A} \times (\nabla \times \mathbf{B}) + \mathbf{B} \times (\nabla \times \mathbf{A}),$$

$$\nabla \times (\mathbf{A} \times \mathbf{B}) = -(\mathbf{A} \cdot \nabla)\mathbf{B} + (\mathbf{B} \cdot \nabla)\mathbf{A} + \mathbf{A}(\nabla \cdot \mathbf{B}) - \mathbf{B}(\nabla \cdot \mathbf{A}).$$

**(10 marks)**

(iv) Using the result in (iii), derive the acceleration formula

$$\frac{D\mathbf{u}}{Dt} = \frac{\partial \mathbf{u}}{\partial t} + \nabla \left( \frac{|\mathbf{u}|^2}{2} \right) + (\nabla \times \mathbf{u}) \times \mathbf{u}.$$

**(5 marks)**

- 2 Consider viscous incompressible fluid on a half plane:  $0 < y < \infty$ , bounded by a solid plate at  $y = 0$ . The fluid is at rest at  $t < 0$  and the plate is jerked into motion at  $t = 0$  and thereafter moving with a constant velocity  $(U, 0)$ . Hence the initial condition for the  $x$ -component of velocity  $u(y, t)$  is

$$u(y, 0) = 0 \text{ for } y > 0$$

and the boundary conditions are

$$u(0, t) = U \text{ for } t > 0,$$

$$u(\infty, t) = 0 \text{ for } t > 0$$

We assume that the flow is two-dimensional  $\mathbf{v} = (u, v)$  with no pressure gradient and that no quantity depends on  $x$ .

- (i) By using the equation of continuity, show that the  $y$ -component of velocity vanishes, that is,  $v = 0$  everywhere. **(3 marks)**
- (ii) Show that  $u(y, t)$  satisfies the equation

$$\frac{\partial u}{\partial t} = \nu \frac{\partial^2 u}{\partial y^2},$$

where  $\nu$  is the kinematic viscosity. **(5 marks)**

- (iii) By introducing the new variables

$$u = Uf(\eta), \quad \eta = \frac{y}{\sqrt{4\nu t}}$$

show that the above equation can be written as

$$\frac{d^2 f}{d\eta^2} + 2\eta \frac{df}{d\eta} = 0.$$

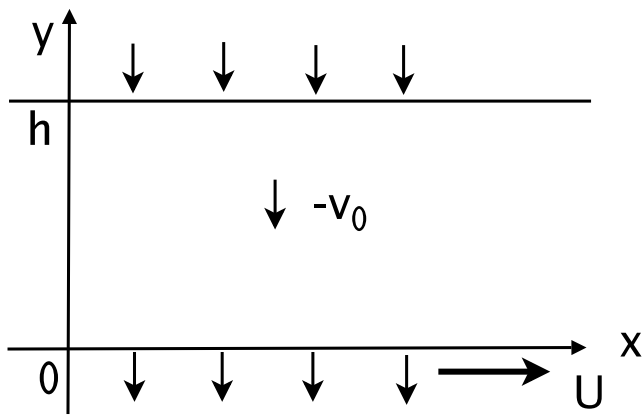
**(9 marks)**

- (iv) State the boundary conditions for  $f(\eta)$ . **(2 marks)**

- (v) Solve the above equation for  $f(\eta)$ .

You may use the error function  $\text{erf}(\eta) = \frac{2}{\sqrt{\pi}} \int_0^\eta e^{-s^2} ds$ , or its equivalent.

**(6 marks)**



- 3 Consider a viscous incompressible fluid of constant density  $\rho$  and of the kinematic viscosity  $\nu$ , between two rigid boundaries  $y = 0$  and  $y = h$ . The lower boundary moves in the  $x$ -direction with a constant speed  $U$  and the upper one is held fixed at rest. The boundaries are assumed to be *porous*, that is, they have tiny holes through which fluid can be sucked in and out.

The vertical velocity of the fluid is given by  $v = -v_0$  (constant) at all time. The problem is two-dimensional, that is, no quantities depend on  $z$ .

- (i) Assuming that the horizontal velocity  $u$  and the pressure  $p$  are functions only of  $y$ , write down the steady (that is, time independent) Navier-Stokes equations for the problem.

(5 marks)

- (ii) Deduce that the pressure  $p$  is constant, that is, independent of the spatial coordinates.

(2 marks)

- (iii) Solve the equation for  $u$  to show that

$$u = U \frac{\exp\left(-\frac{v_0 y}{\nu}\right) - \exp\left(-\frac{v_0 h}{\nu}\right)}{1 - \exp\left(-\frac{v_0 h}{\nu}\right)}.$$

(10 marks)

- (iv) Deduce an approximate formula for velocity which is valid close to the lower boundary, when  $v_0 h / \nu$  is large in comparison to 1. Sketch the velocity profile therein.

(5 marks)

- (v) From the expression  $u$  in part (iii), derive the velocity  $u$  for the limiting case of  $v_0 \rightarrow 0$ .

(3 marks)

4 Consider a two-dimensional, incompressible viscous fluid in the region between two infinite horizontal plates along  $y = 0$  and  $y = h$ , flowing steadily in the  $x$ -direction.

(i) Assuming that  $\mathbf{v} = (u(x, y), 0)$  show that  $u$  is actually a function of  $y$  only. **(2 marks)**

(ii) Write down the components of the Navier-Stokes equations by neglecting external body forces and show that the pressure gradient is constant and acts only in the  $x$ -direction. **(12 marks)**

(iii) If the plates are fixed and  $\frac{dp}{dx} = -P$ , show that

$$\mathbf{v} = \frac{-P}{2\mu}y(y - h)\mathbf{e}_1,$$

where  $\mathbf{e}_1$  is a unit vector in the  $x$ -direction and  $\mu$  denotes the viscosity.

**(5 marks)**

(iv) Calculate the volume flux per unit area across any fixed plane perpendicular to the motion. **(3 marks)**

(v) Calculate the drag per unit area exerted by the fluid on the plane  $y = 0$ . **(3 marks)**

- 5 (i) A fluid moves in a steady two-dimensional flow in the region defined by  $x \geq 0$ ,  $y \geq 0$ . The boundary with equation  $y = 0$  is occupied by a stationary flat plate. Given that the  $x$ -component of velocity  $u \rightarrow U$  as  $y \rightarrow \infty$ , where  $U$  is a constant, write down the expressions for
- (a) the displacement thickness  $\delta_1$  of the boundary layer, and
- (b) the momentum thickness  $\delta_2$  of the boundary layer.

(4 marks)

(c) When the flow is given approximately by

$$\frac{u}{U} = \left(\frac{y}{\delta}\right)^m \quad \text{where } 0 < m < 1,$$

compute  $\delta_1$  and  $\delta_2$  explicitly and state which is the largest.

(6 marks)

- (ii) Consider a steady two-dimensional flow past a semi-infinite solid boundary along  $y = 0$  in the region  $y \geq 0$ ,  $x \geq 0$ . Blasius's boundary layer equations for it are given by

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = \nu \frac{\partial^2 u}{\partial y^2},$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0,$$

where  $\nu$  is the kinematic viscosity. Derive the energy equation for the boundary layer

$$\frac{d}{dx} \int_0^\infty \frac{u}{U} \left(1 - \frac{u^2}{U^2}\right) dy = \frac{2D}{\rho U^3},$$

where  $u$ ,  $v$  are the  $x$ - and  $y$ - components of the velocity and

$$D = \mu \int_0^\infty \left(\frac{\partial u}{\partial y}\right)^2 dy$$

is the dissipation rate of energy. The constant  $U$  is the  $x$ -component of the velocity as  $y \rightarrow \infty$ .

(15 marks)

**End of Question Paper**